U.S. Gov t

SAA09VE14-001

REV. E

B/L: 006.01 SYS: ORBITER

ACCESS ARM

NOV 1 2 1992

Critical Item:

UPPER HINGE ACTUATOR (2 Items Total)

Find Number:

A101349

Criticality Category:

15

SAA No:

09VE14-001

System/Area:

ORBITER ACCESS ARM /

LC-39 PAD A&B

NASA

PMN/

U70-0503

Name:

ORBITER ACCESS ARM

Mfa/

Part No:

75M12560 FLOTORK

Drawing/

79K06013 , 79K40015

Part No:

NONE

Sheet No:

Function: Extends and retracts the Orbiter Access Arm. Upper actuator assembly (A101349) is redundant with Lower Actuator Assembly (A101364).

Critical Fallure Mode/Fallure Mode No: Fails to actuate / 00VE14 001.001

Failure Cause: Seizes, severe leakage.

Failure Effect: Arm cannot be extended or may exceed the 30 second maximum allowable time due to mechanical seizure or severe leakage. Possible loss of life.

ACCEPTANCE RATIONALE

Design:

- Four (4) piston actuators coupled to the hinge post weldment act in a redundant manner.
- Seal design (square cross-section) minimizes tendency to rolt and .035" piaton land clearance reduces princibility of seal causing acizure.
- Combination of bronze liner and steel piston extremely reliable with respect to seizure.

 Emergency Extend and Retract test is performed prior to each launch, File VI OMRSD requirement.

Inspection:

Actuator inspected for external leakage annually per OMI V6C50.

Fallure History.

 The PRACA database was researched and no fallure data was found on this component in the critical failure mode.

WORKSHEET 5312-013 920705ggPS0128

Attachment 5050234AG

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• The GIDEP failure data interchange system was researched and no failure data was found on this component in the critical failure mode.

Operational Use:

• Correcting Action:

There is no action which can be taken to mitigate the failure effect.

• Timeframe:

Since no correcting action is available, timeframe does not apply.

Attachment 905023446 Sheet 3 of 8